

1 **Title: Material selection and detailed design of optical windows under high loads**

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6

7 **Abstract**

8 In this study, the design aspects of optically accessible pressure vessels are investigated via a case study of a High Pressure
9 Combustor experimental rig. The rig was designed to take optical measurements of combustion, simulating the conditions
10 found in internal combustion engines and turbines. Although, it is not new to equip chambers and reactors with sight
11 windows, important aspects of design and relevant information regarding optical access is missing or are insufficiently
12 explored or not readily accessible in the existing literature. A comprehensive review of requirements for optical access to
13 such high-pressure, high-temperature systems has been conducted. It is shown in a readily-navigable format as function of
14 application and precision, with data and technical correlations hitherto not found in a 'user-friendly' style. The material
15 selection procedure is detailed and supported by a complete comparison of optical materials and relevant properties. The
16 review revealed a significant inconsistency in mechanical properties claimed in the literature for optical materials. As a
17 response to this, increased safety factor values are suggested as function of level of uncertainties and effects of failure,
18 typically three to four times higher than the industrial standard. Moreover, newly developed equations are presented
19 linking performance analysis to the design criteria.

20

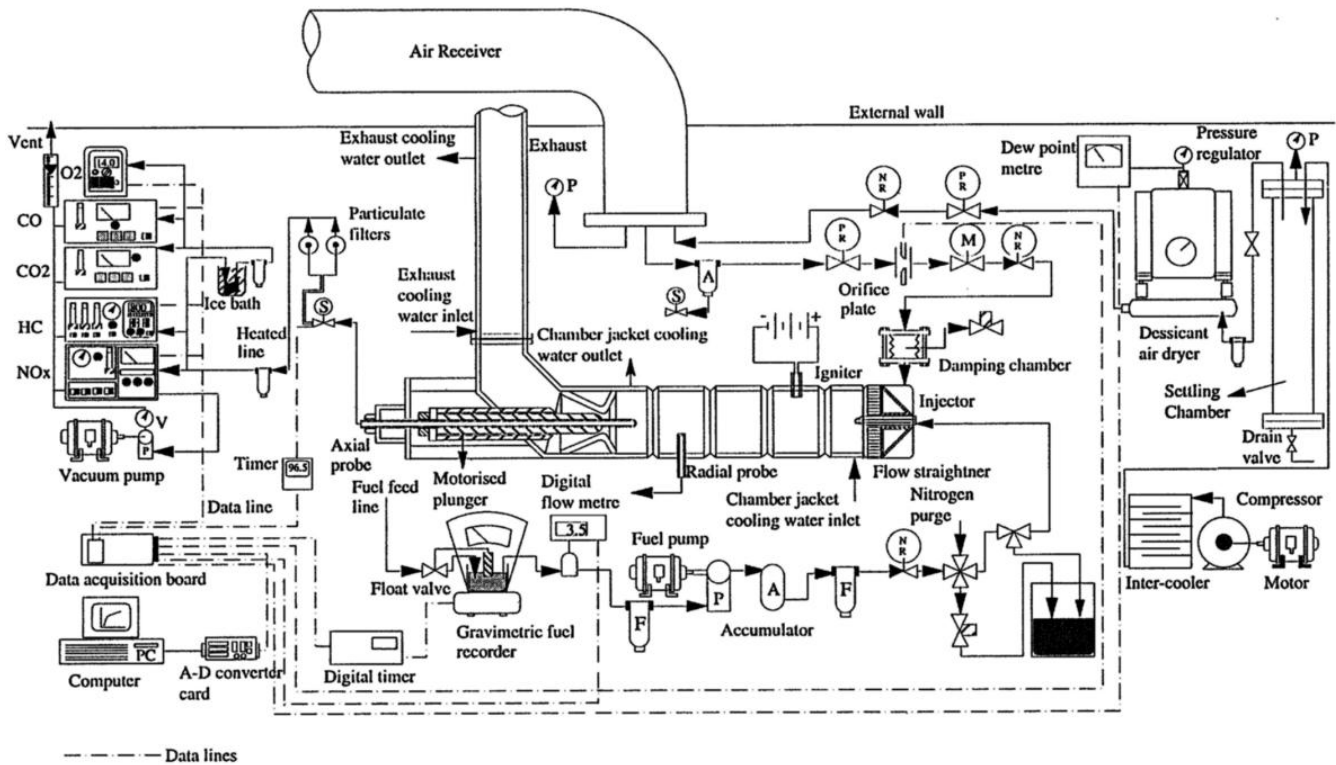
21 Keywords:

- 22 • Optically-accessible reactor,
- 23 • Optical engine,
- 24 • Pressure vessel;
- 25 • Sapphire,
- 26 • Window design,
- 27 • Combustion

28 1 Introduction to the High Pressure Combustor (HPC) and the need for optical development

29 Over the past two decades, concerns about global warming and the depletion of the ozone layer have driven researchers
30 to find better alternatives to the high energy consumption demand [1-4]. With the combustion of fossil fuel and the
31 subsequent production of carbon dioxide being accounted for as the main contributor to the current release of
32 greenhouse gases to the atmosphere [5], and taking into consideration that a solution to the current energy supply
33 problems is yet distant, improvements in the understanding of the chemical reaction and flame-propagation processes
34 and reduction the emissions of these engine-fuel combinations should be implemented as a short term solution [6-8].

35 The HPC was developed to address research topics in combustion science. Its unique design makes it a versatile tool to
36 model and test the working, at real-life conditions of industrial furnaces, external and internal combustion engines and gas
37 turbines. It can be set up to test steady combustion up to 60 bar for 30 minutes. It can accept virtually any combustible
38 substance with a high accuracy of air-fuel ratio and a control of residence time. Moreover, the flow pattern can be set to
39 either plug or swirl. The high-pressure air (variable up to 60 bar) is delivered to the chamber via a number of safety
40 instruments from a large air receiver – which is charged by a three-stage piston pump. The air arriving to the combustion
41 chamber is dried, and its flow and pressure is set by a computer-controlled valve system. The fuel is injected into the
42 chamber by interchangeable injectors; the fuel flow pattern, supply pressure and volume flow is variable. The actual
43 combustion chamber is not a single-piece vessel but rather an assembly of several sections. Therefore, the length of the
44 chamber – and hence, the residence time of reactants – can be varied depending on the application. The sections were
45 designed with numerous radial access points so that reaching any point inside the combustor for sampling would be
46 possible; see Figure 1 for a schematic of the experimental rig. The initial ignition is provided by a high-energy spark. At the
47 end of the process, the burned mixture leaves the chamber via a special plunger valve that is capable of withstanding the
48 high temperature and pressure. A detailed description of the HPC facility can be found in [9-11].



49

50

Figure 1. Schematic drawing of the experimental rig, adapted from [10].

51 The HPC has supplied the combustion research community with vital information. Using thermocouples and sample
52 probes, its current capabilities have been fully utilised. New developments were needed to keep the rig up-to-date. In the
53 last 15 years or so, the optical- and laser-based measurements became the most important tools to investigate
54 combustion details and results of these methods published elsewhere [12-14]. These methods have a fast response time
55 and do not require actual physical contact with the flames. It was essential to equip the HPC with optical access in order to
56 keep the research work current and on-going.

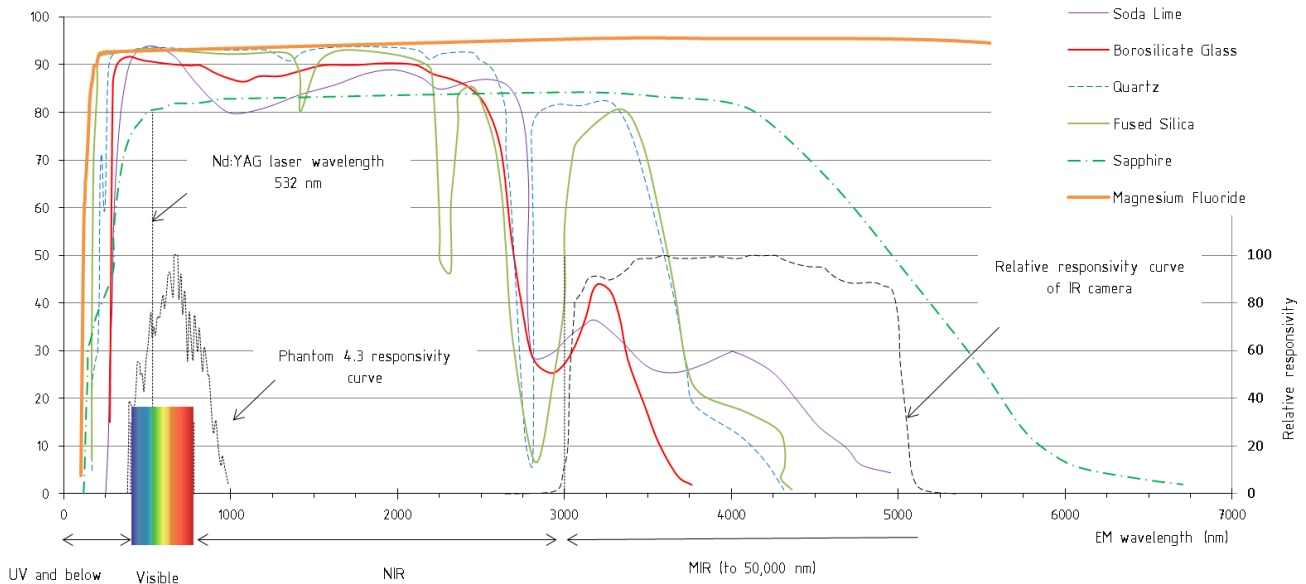
57 2 Current contribution

58 The design methods and procedures of industrial pressure vessels are well-documented, with comparisons of methods
59 and standards available for industrial applications [15-25]. These provide good guidelines even for an unconventional
60 design task, but of course they do not provide comprehensive data for all possible cases.

61 In general, the available literature lacks data regarding optical access to pressurised vessels. Information on design
 62 practices and material properties are scattered in the literature, being hard to find and often inconsistent. In this work,
 63 novel complementary material is provided for the design of chambers and reactors that require the equipment of sight
 64 windows on them. As a result of extensive review, the properties of practical optical materials were collected and
 65 presented together in graphs and tables, allowing for direct comparison. The details of special design practices regarding
 66 transparent parts are discussed, and the available data on existing design solutions is collected and shown. Some
 67 complementary material is added to the basic equations and relations in Statics. Moreover, there are papers examining
 68 design procedures [26], but to the authors' knowledge this is the first detailed design study on optically accessible
 69 pressure vessels (fixed volume or internal combustion optical engine) where the real-life application of the collected data
 70 is shown. Structural analysis of optical material window is shown and its effects to practical design.

71 3 The optical access: review of material and their properties; practical solutions; 72 mechanical and optical performance

73 3.1 Optical materials: mechanical, optical and chemical properties



74
 75 **Figure 2 Transmittance of the reviewed optical materials; relative responsivity and wavelength of interest are also**
 76 **shown (for Design requirements section).**

77 There are a large number of materials that can be considered for sight windows on pressure vessels, from ordinary plastics
 78 to exotic ceramics. In this work, only the most common and most practical optical materials were chosen for comparison.

79 The underlying design criteria for selecting the optimal optical material type are: useful transmittance range, operating
 80 temperature and mechanical load. In Table 1, only high-operating-temperature materials are listed. It is important to note
 81 there are other choices available for specialised tasks, such as silicon or germanium, but their availability is limited and
 82 they are costlier. In low-temperature environments, plastics like acrylic and polycarbonate can be used. During design, it is
 83 essential to consider the working temperature and obtain a good estimate of it from simulations or experiments.

84 Although unusual in mechanical engineering, it is important to choose the right material for the required electromagnetic
 85 band. It is also vital to consider the ratio of the electromagnetic energy falling on a body to that transmitted through it.
 86 This ratio is called the transmittance of the material [27]. Transmittance values for each wavelength vary significantly
 87 among material. For example, a larger selection of materials can be considered if the investigated radiation is in the visible
 88 or in the near infra-red (NIR) regions. Due to the availability of a wider range of materials, the implication is that sight
 89 windows for high speed imaging or laser-aided measurements can be designed more easily, and more complex shapes
 90 with larger dimensions are therefore possible. Choosing an optimal material is more complex when longer wavelengths
 91 have to be captured for both spectroscopy and thermal imaging. For wavelengths over 2500 nm, the transmittance curves
 92 start fluctuating or becoming discontinuous. If this, then, is the electromagnetic wave band region of interest, careful
 93 planning will be needed to select the right material type. The transmittance of common optical materials for wavelengths
 94 under 200-250 nm falls rapidly. Yet, it is an important region in combustion science as some radicals have their peak
 95 emissivity in this electromagnetic band. Researchers and designers are practically left with fused silica and a number of
 96 fluorides (MgF_2 , CaF_2 , BaF_2) to use. Figure 2 shows the transmittance curves of selected materials.

97 Once the material candidates are shortlisted by wavelength transmittance, the more conventional design process follows
 98 this when further mechanical, thermal and chemical resistance properties are of interest.

99 Finally, the cost analysis needs to be taken into account when the material type providing the optimal solution is chosen.

100 Table 1 summarises some of the most related properties of a selection of practical optical materials. As expected, all of the
 101 listed properties are functions of temperature, size and shape, exact composition, heat treatment, surface finish, and
 102 other manufacturing processes. It is important to note that there are significant differences (10-15%) between the claimed
 103 values by different manufacturers and textbooks.

104

Table 1. Optical material properties

	Unit	Soda Lime Glass	Borosilicate	Quartz	Fused Silica	Sapphire	Magnesium Fluoride
General							
Chemical Formula, Composition	(weight %)	SiO ₂ :74, Na ₂ O:15, CaO:5, others	SiO ₂ :80+, B ₂ O ₃ :7-13%, Na ₂ O, others	SiO ₂ :99	SiO ₂ :99	Al ₂ O ₃ :99	MgF ₂ :99
Density	(g/cm ³)	2.2-2.52	2.2-2.4	2.2	2.2	3.98	3.18
Optical							
Useful Transmission	(nm)	320-2300	325-2100	200-2400	180-2200	150-5000	110-7500
Refractive index (588 nm)	-	1.52	1.47	1.46	1.46	1.76	1.38
Mechanical^I							
Young's Modulus	(GPa)	72	64	73	73	335	138
Tensile Strength	(MPa)	41	27-62	50	50	275 ^{II}	140
Hardness, Vickers	-	550	520-580	1000-1200	1000-1200	1940	400
Poisson's ratio	-	0.23	0.21	0.17	0.17	0.25	0.27
Weibull variability of strength	-	6 ^{III}	30 ^{IV}	8.82 ^V	10.2 ^{VI}	5	5
Weibull stress	(MPa)	129 ^{III}	71 ^{IV}	115 ^V	180 ^{VI}	485	96
Thermal							
Softening Point	(°C)	1450	800-850	1730	1600	2300 ^{VII}	1255
Max. Continuous Operating Temperature	(°C)	260	280-350	950-1150	950-1100	1200	500
Thermal Conductivity at 300 K	(W/mK)	0.96	1.1-1.2	1.38	1.38	27.21	11.6
Coefficient of Expansion	(10 ⁻⁶ /K)	3.5-9	3.25-4	0.55	0.55	8.4	8.9

properties perpendicular to optical axis

materials are birefringent for exact refractive indexes see references

^I mechanical and optical properties are dependent on fabrication method and surface finish; ^{II} fractural strength; ^{III} Kimble R-6; ^{IV} BK-7; ^V standard polish; ^{VI} "super polish"; ^{VII} melting point

105

106 Soda lime glass is the common glass type that can be found everywhere. It is mass-manufactured by floating the hot raw
 107 material on a bed of molten tin. It is the least expensive material of all, and being softer than other glasses, it is easy to
 108 make a complex part out of it. It is a hard material with good scratch resistance, but is significantly softer than other
 109 glasses or sapphire. It is not resistant to many chemicals, and its higher coefficient of expansion makes it sensitive to
 110 uneven temperature distribution [28-32].

111 Borosilicate glass is 2-3 times more expensive than soda lime glass but still considerably less expensive than fused quartz
 112 or silica. It has the same easy manufacturing properties as soda lime but usually has a lower thermal expansion coefficient,
 113 hence making it more resistant to thermal shock. Leaching can occur but it is more resistant to chemicals [28-31, 33, 34].

114 Fused quartz and silica have very similar properties as they have an almost identical composition. The main difference
 115 between them is in the amount of contamination caused by the different manufacturing processes. Quartz is made from
 116 melted and cleansed naturally occurring quartz sand with larger amount of contamination in the product, while fused
 117 silica is a pure version of quartz synthesised from various gases. However, their mechanical and electrical properties are
 118 identical. The only contrasting (and significant) advantage is that, silica has an excellent transmittance in the ultra violet

119 (UV) region. This property makes it unique among silicon oxides. A major advantage of quartz and silica, when compared
120 to cheaper glasses, is their increased stability. Their mechanical properties are significantly less sensitive to temperature
121 changes than borosilicate or float glasses. For instance, for a borosilicate, the linear thermal expansion at 500 °C increases
122 its ambient value a few hundred times; silica, however, faces an increase of about 40 times and then stays constant with
123 further increase of the temperature. This makes the evaluation of thermal stresses a lot easier when implementing quartz
124 and silica. Nevertheless, their excellent properties come at a price: the material cost is significantly higher than the
125 aforementioned glasses and their higher temperature resistance makes fabrication more complex. They have a reasonably
126 good resistance to chemicals but break down with some caustics, fluorinated acids and plasmas [28-31, 33-36].

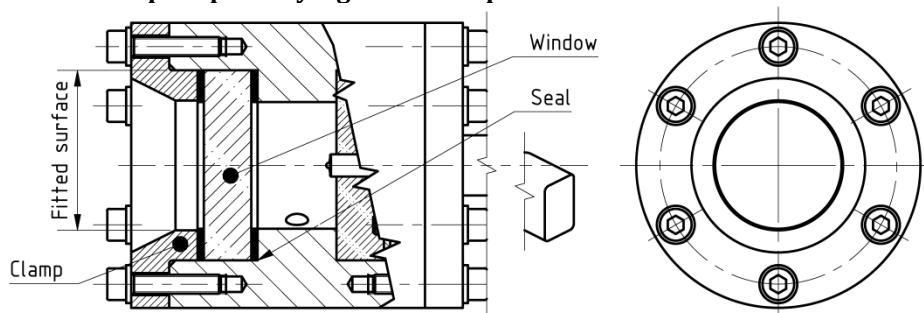
127 Sapphire is a single crystal and a very versatile material. It is the second hardest material on Earth, which makes it best
128 choice of material whenever wear and abrasion are the main constraints. Its high mechanical strength and modulus of
129 elasticity provides good resistance against impacts. It is virtually impervious to all corrosive materials and its thermal
130 stability outperforms all other optical materials. Yet, sapphire raw material is not significantly more expensive than fused
131 silica. On the other hand, its extreme hardness and a high melting point make the manufacturing process challenging and
132 costly. In conclusion, sapphire is not suitable for large windows and for complex shapes [37-43].

133 Magnesium fluoride is an excellent material choice for application in the UV bandwidth (the cheaper CaF2 has similar
134 properties but with slightly reduced useful transmittance range). Larger size crystals can be grown, and it is possible to
135 machine it with standard diamond tools as this material can be polished well. Thus, complex shapes and geometries can
136 be achieved. It has a wide range of transmissivity but it is not as wear-resistant as the other materials, and its surface will
137 degrade in a humid environment at elevated temperatures (over 500 °C) [33, 39, 44-46].

138 3.2 Mounting methods

139 There are a number of different ways to hold the optical element within an optical apparatus. A particular mounting
140 method can be selected considering the geometric constraints, the sealing requirements, position accuracy, the
141 orientation of optical axis, stress and the deformation caused by pressure difference, and birefringence. In this paper,
142 sight optics is investigated only; their mounts are less complex than lenses that need more degrees of freedom.

143 3.2.1 Optical element kept in place by a guided clamp



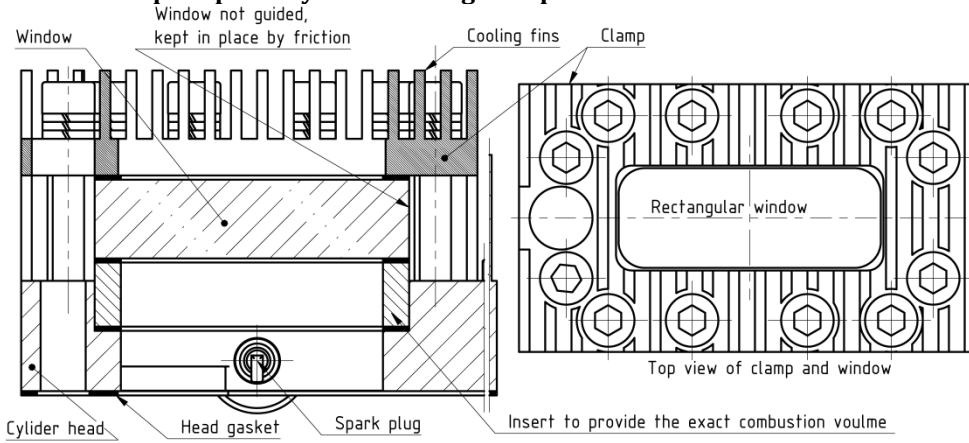
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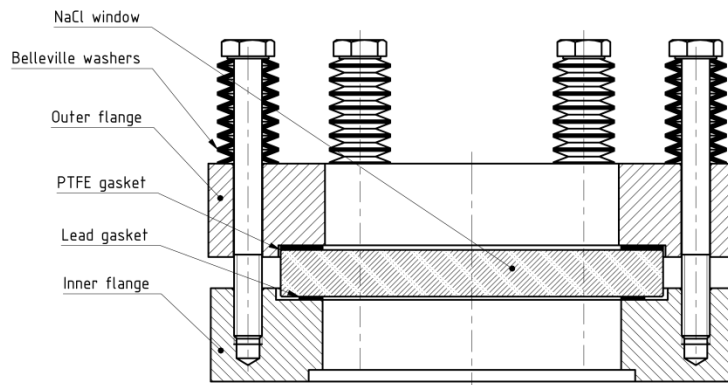
Figure 3. Fixed volume combustion chamber with circular window that is positioned by a guided clamp [47].

146 Figure 1 shows the usual clamping method where the retainer is fitted and guided in the direction of the displacement of
147 the window. The radial position of the retainer is fully defined by the contact forces. The advantage of this solution is the
148 simple tensile load on fixing bolts, and simplified dismantling and re-assembly. Details of loaded bolted joints can be found
149 in the literature [48]. The disadvantage here is that the larger the size in the direction of optical axis, the more complex its
150 design and manufacturing turns out to be [49].

151 **3.2.2 Optical element kept in place by a free sitting clamp**



152
153 **Figure 4. Four-stroke optical engine, the rectangular window is sandwiched by the clamp and soft gaskets [6, 7].**



154
155 **Figure 5. Special sodium chloride free sitting window for a high-temperature, high pressure difference, IR spectra [50]**

156 The simplest design solution is illustrated in Figure 4 and Figure 5. The clamp is not guided but constrained by contact in
157 one axis and constrained by friction along the other two axes; its position is defined by fixing bolts. Its advantages are: a
158 simpler design, easier to manufacture, smaller in size along the optical axis, and that its position along the optical axis can
159 easily be varied. Its disadvantage is that a greater amount of mechanical (bending) load on bolts is required; since the
160 window can freely move, bringing the assembly together can also be problematic.

161 **3.2.3 Adhesives**

162 Fixing an optical element in a carrier frame using adhesives, as indicated in Figure 6 is a convenient solution for lower
163 pressure and temperature environments. In both cases, the window sits against a shoulder which provides an accurate
164 positioning. All mechanical loads rising from the pressure differential are taken by the adhesive. In the second case, the
165 adhesive acts as a sealant and retainer; only, the stress is induced, but the pressure difference is taken by the shoulder on
166 the frame cell. The main advantage of this solution is the modest space requirement. Its only disadvantage is that the
167 performance of the assembly is proportionately dependent on the properties of the adhesive, which are usually limited.

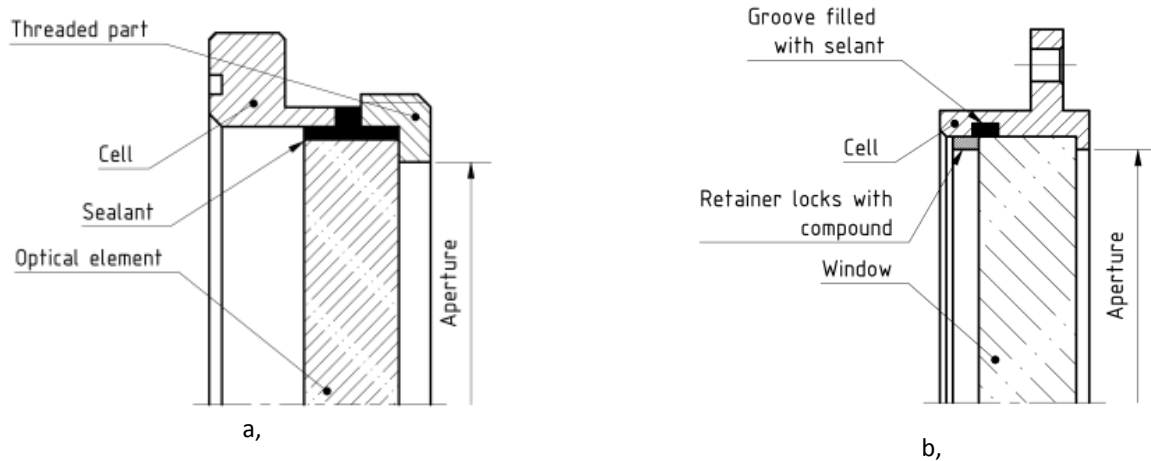
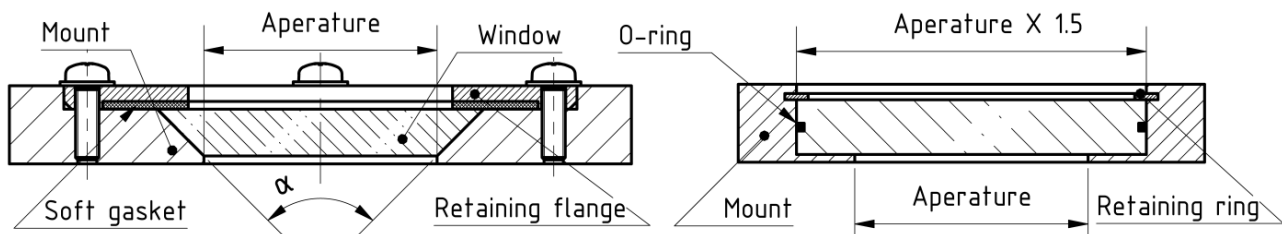


Figure 6. a, Using adhesives for a low pressure application [51]; b, Using adhesives for a moderate pressure application [52].

169

170 **3.2.4 Fitted inside the shell of the vessel**



171

172

Figure 7. Window integrated in the vessel body [53, 54].

173 The optical element can be fitted inside the housing, a typical application area is the deep submergence vehicles, Figure 7.
 174 There is no need for bolts as the major load-bearing element. Its advantage is that this setup can take large pressure
 175 differences, while the vessel geometry can also be made more simply. Its disadvantage is that the window can only be
 176 dismantled from the pressurised side, more complex window geometry required.

177 **3.3 Mechanical performance**

178 **3.3.1 Allowable or design stress in the optical element, safety factor**

179 The estimation of the allowable or design stress is among the most important and sometimes challenging tasks, especially
 180 at elevated temperatures [23-25, 55-57]. The data of mechanical properties can be found in the literature for the more
 181 common materials; however, there often is no consistency in the given values. It becomes even more difficult to find
 182 information when practical issues are being considered, such as the effects of temperature, humidity, manufacturing
 183 technology, surface finish, and loading rate. Pressure vessel codes provide suggestions for high strength alloys which can
 184 then be taken as a first guidance for optical materials. According to BS EN 13445-3 [56] and ASME Boiler and Pressure
 185 Vessel Code Section VIII [57], the design stress should be calculated as:

(EN)
$$\sigma_{des} = \min\left(\frac{R_{p0,2/T}}{SF}; \frac{R_m/20}{SF}\right) = \min\left(\frac{R_{p0,2/T}}{1,5}; \frac{R_m/20}{2,4}\right) \quad (1)$$

(ASME)
$$\sigma_{des} = \min\left(\frac{R_{p0,2/T}}{SF}; \frac{R_m/20}{SF}\right) = \min\left(\frac{R_{p0,2/T}}{1,5}; \frac{R_m/20}{2,14}\right) \quad (2)$$

186 where, σ_{des} is the allowable design stress; SF is the safety factor; $R_{p0,2/t}$ is the 0,2% proof strength at T temperature; R_m
 187 is the tensile strength at 20 °C [58]. As optical materials discussed in this work have brittle characteristics, it is only the
 188 safety factors that are associated with the tensile strength that are applicable. It is suggested that the safety factor for
 189 optical design should always exceed 2. The general value for a well-designed system is around 3, when failure is not
 190 expected to cause major damage. When there is more uncertainty in the design, the usual and conservative safety factor
 191 value is 4. The value can be as high as 5 for non-optimum or unplanned conditions (manufacturing or usage) or when
 192 failure can cause significant damage [51, 59].

193 **3.3.2 Geometric and mechanical tolerances**

194 The tolerances on diameters and on the thickness of the centre and edges are comparable to general precision
 195 manufacturing, typically h6 to h11. When the edge of the window is not fitted and/or it is not a sealing surface the size

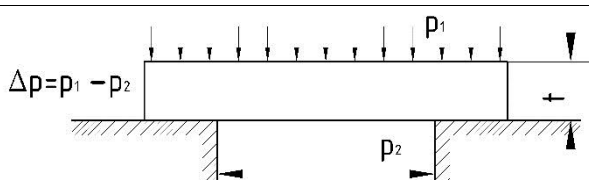
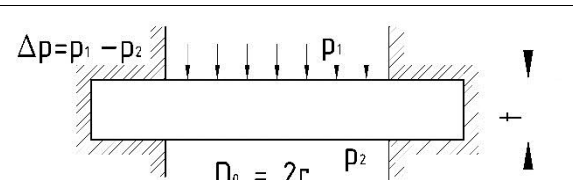
196 tolerance is in the 0.1-0.01 mm range. The tolerance on the thickness of the optical access has importance for lenses but
 197 for windows it is not crucial. Similarly, the usual parallelism requirements are on the fine side but are comparable but to
 198 the ones used in pressure vessel manufacturing. However, the surface roughness values are in a couple of order less than
 199 the typical values for precision manufacturing technologies for metals. Moreover, the quality of the finish is further
 200 described by the scratch and dig number. A usual scratch/dig specification consists of two numbers e.g. 80/50. The first
 201 number indicates the maximum size of cracks (scratches) on the surface of the optical element. The second number
 202 describes the maximum size of round-shaped imperfections: digs and pits [60, 61].

203 Guidelines are given in Table 3 for selecting optical and mechanical properties for sight window applications.

204 3.3.3 Deflection and stress

205 Equations relating deflection to the applied pressure difference can be found in the literature for a number of shapes and
 206 support modes [62, 63]. In this work only the details of relations for the plane-parallel circular window shape are shown.

Table 2, Classical mechanics of plane-parallel circular elements

Non supported case	Supported case
	
$\delta = \frac{3(3 + \mu) \Delta p r^2}{8 t^2} = K_w \frac{\Delta p r^2}{t^2} \quad (3)$	$\delta = \frac{3 \Delta p r^2}{4 t^2} = K_w \frac{\Delta p r^2}{t^2} \quad (4)$
$K_w = 0.125 \quad (5)$	$K_w = 0.75 \quad (6)$
$x = \frac{3}{16} (-\mu^2 - 4\mu + 5) \frac{\Delta p r^4}{E t^3} = K_x \frac{\Delta p r^4}{E t^3} \quad (7)$	$x = \frac{3}{16} (1 - \mu^2) \frac{\Delta p r^4}{E t^3} = K_x \frac{\Delta p r^4}{E t^3} \quad (8)$
$K_x = 0.96 \quad (9)$	$K_x = 0.17 \quad (10)$

207

208 K_w is a generalised constant suggested by textbooks [51, 52, 59]. In these works, K_w is chosen conservatively to cover a
 209 wide range of optical materials. This conservative method was chosen in this study to make a suggestion for values of K_x .
 210 In the rest of the equations, (3)-(10) δ is the stress; μ is the Poisson ratio; Δp is the pressure differential; r is the radius
 211 which is half of the aperture or diameter D_0 ; x is the deflection; t is the thickness of the optical element; E is Young's
 212 modulus. If the stress equations are rearranged and the safety factor, the diameter and design stress are inserted,
 213 then the minimum required thickness of the optical element can be calculated.

$$t_{min} = \left(\frac{1}{2} D_0 \right) \left[\frac{K_w S F_\delta \Delta p}{\delta_{des}} \right]^{1/2} \quad (11)$$

214 where, t_{min} is the minimum thickness of the circular optical element; D_0 is the diameter of the aperture; $S F_\delta$ is the safety
 215 factor; Δp is the applied pressure difference on the optical element; δ_{des} is the allowable design stress. Using Equations (3)
 216 and (8), the deflection can be calculated or the rearranged version with the maximum allowable deflection can be used
 217 to find the minimum required thickness:

$$t_{min} = \left[\frac{S F_x K_x \Delta p D_0^4}{16 E x_{max}} \right]^{1/3} \quad (12)$$

218 where $S F_x$ is the safety factor. In general, as $S F_\delta$ is associated with complete breakdown and failure and $S F_x$ has an effect
 219 on only the quality of the image produced by the optical element. $S F_x$ can have a significantly lower value than the $S F_\delta$.
 220 Equation (12) provides results for a simple case of a mechanical load. When, there is a combined load from thermal and
 221 mechanical loads, the deflection needs to be calculated using Finite Element Analysis (FEA). The result of the simulation
 222 can be used to calculate the outer radius (R) of the window that can be turned into a divergent meniscus lens (assuming
 223 the same deflection on both sides of the window):

$$R = \frac{x^2 + D_0^2}{8x} \quad (13)$$

224 Then with the known thickness the lens power (P_{lens}) can be calculated:

$$P_{lens} = (n - 1) \frac{-t}{R^2 - Rt} \quad (14)$$

225 The maximum deflection of a window is a function of allowable image distortion. In an optical system with lens and
 226 sensor, the lens focusing error usually gives the tolerance in dioptres. It is hard to find tolerances published, but as a rule
 227 of thumb some values are summarised in Table 3, [64-67].

228 Equations of stress, deflection and power calculation for rectangular, plan parallel windows can be found in [68].

229 3.3.4 Failure estimation by statistical tool

230 It is a common practice to implement Weibull statistics to estimate the probability of failure (P_f) when a given σ load is
 231 applied on a brittle material.

$$P_f = 1 - \exp \left[- \left(\frac{\sigma}{\sigma_0} \right)^m \right] \quad (15)$$

232 where m is a constant describing the variability in strength; their values having been experimentally determined and
 233 published. σ_0 is a stress level at which 63% of the samples fail, m is the so called Weibull modulus and indicates the scatter
 234 of fracture stress around σ_0 [51, 69-72]. The acceptable values can differ significantly and they should be determined for
 235 each application individually. Some suggested examples: for a cheap easily replaceable cutting tool - 10^{-2} ; for an expensive
 236 part that upon failure can cause serious damage - 10^{-4} ; when personal injury is at risk - 10^{-6} ; when the outcome of a
 237 failure could be fatal then 10^{-8} .

238 3.4 Optical performance

239 3.4.1 Birefringence and maximum optical path difference (OPD)

240 It is usual for most practical optical materials to have two indices of refractions. Their refractive index is a function of the
 241 propagation-direction and polarisation of the incident electro-magnetic wave. Furthermore, it is a function of the
 242 mechanical stress in the medium. Optical substances having this property are called birefringent materials [73]. The level
 243 of birefringence is expressed as a difference in the optical path of two perpendicular states of the polarised wave. This
 244 inequality in distance is called the OPD and it is measured in nanometres. The OPD has been previously investigated for
 245 plane-parallel circular plates with a pressure differential applied on them; Sparks et al. [74] derived an approximate
 246 relation:

$$OPD = 8.89 \times 10^{-3} (n - 1) \frac{\Delta p^2 D^6}{E^2 t^5} \quad (16)$$

247 where, OPD is the optical path difference; n is the refractive index of the material; Δp is the pressure difference applied
 248 across the planes of the optical element; D is the aperture, the unsupported diameter of the optical element; E is Young's
 249 modulus of the medium; and t is the thickness of the window. This OPD caused by an applied stress called the stress
 250 birefringence. It is measured as OPD per unit travel path; its unit is nm/cm. The details of the maximum allowable
 251 tolerances on birefringence for some applications are given in ISO 10110-8 [61] and Kimmel and Parks [75]; a summary is
 252 presented in Table 1. Equation (16) can be rearranged to find the minimal required thickness:

$$t_{min|OPD\Delta p} = \sqrt[5]{8.89 \times 10^{-3} (n - 1) \frac{\Delta p^2 D^6}{OPD \cdot E^2}} \quad (17)$$

253

Table 3. Sight window properties for different applications

Precision	Typical application	Maximum power of a deflected window (diopetre)	Maximum OPD per unit path length (nm/cm)	Parallelism or plane angle (degree)	Flatness (λ is the characteristic wavelength)	Surface finish or roughness (nm)	Surface finish quality
Extreme	Polarisation and interference instrumentation, deep-space instrumentation	No data	2	No data	$\frac{\lambda}{20}$ or better	0.3	No data
High	Photolithography optics and astronomical telescopes	No data	5	0.001	$\frac{\lambda}{10}$	0.5	10/5
Good	Photographic and microscope optics, visual telescope	$10^{-2} - 10^{-6}$ *	10	0.01-0.001	$\frac{\lambda}{2} - \frac{\lambda}{4}$	1	40/20-20/10
Semi	Eyepieces, viewfinders, magnifying glasses	10^{-1}	20	0.1-0.01	λ	2	60/40
Commercial	Illumination optics, condenser lenses	No req.**	No req.	0.1	No req. - 2λ	4	80/50

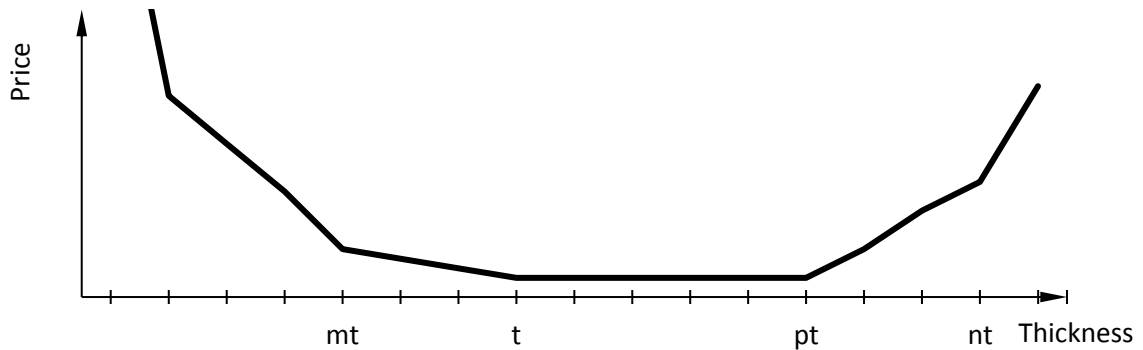
*in general it can be said that, tolerance values in the order of 10^{-6} or less are likely to be negligible when they are compared to the uncertainty in the focus adjustment of a lens system
 ** No req.: No requirement

254 It is important to note that there are always some residual stresses in optical materials, depending on the quality of the
 255 manufacturing processes. More details relating the manufacturing process to stress birefringence can be found in the
 256 references.

257 **3.5 Factors limiting the maximum thickness**

- 258
- 259 • The most obvious limiting factor is the available space the geometric constraints, which depends on the individual
 260 design. The different possible mounting methods and previous publications of solutions are introduced in other
 261 sections of this work.
 - 262 • Transmittance change as a function of material thickness [27]. Significant decrease of the transmittance can only
 263 occur with large thicknesses, this is not a usual design constraint for high load applications.
 - 264 • Temperature gradients can cause stress concentration in window materials. For heated or cooled designs this can
 265 limit the size of the geometry. Ceramics with larger thermal conductivity coefficients are less sensitive to thermal
 266 shock, [76, 77].
 - 267 • Price: manufacturing and material cost
 268 As Figure 8 indicates typically there is a thickness range where the price is at its minimum. This is the most
 269 commonly made size range ($t \leftrightarrow pt; p \sim 3 \dots 5$) that is mass produced with a variety of tolerances and finishes.
 270 These are usually used in general optics and not adequate for high load applications. To the left from this region
 271 ($t \leftrightarrow mt; m \sim 0.6 \dots 1.0$) there is an increase in price where the manufacturing becomes more laborious. The
 272 relatively thin geometry makes the window fragile and prone to deflection under the manufacturing loads. A
 273 further sharp rise expected for thicknesses below mt where extra care is required to provide precision finish. For
 274 example, during manufacturing there is a 50% breakage rate for 0.2mm borosilicate glass coverslips. On the right
 hand side from the flat ($pt \leftrightarrow nt; n \sim \text{few hundreds}$) the increase is driven by the cost of material. nt represents

275 the maximum size that is achievable using the standard or already existing raw material production tooling.
276 Larger geometries can only be made if tooling cost is covered.
277 Figure 8 only introduces general trends in the price the actual values will differ from geographic region to region,
278 material type, and quantity required.



279

280 **Figure 8. Approximate cost of manufacturing of disc-shaped windows as function of thickness**

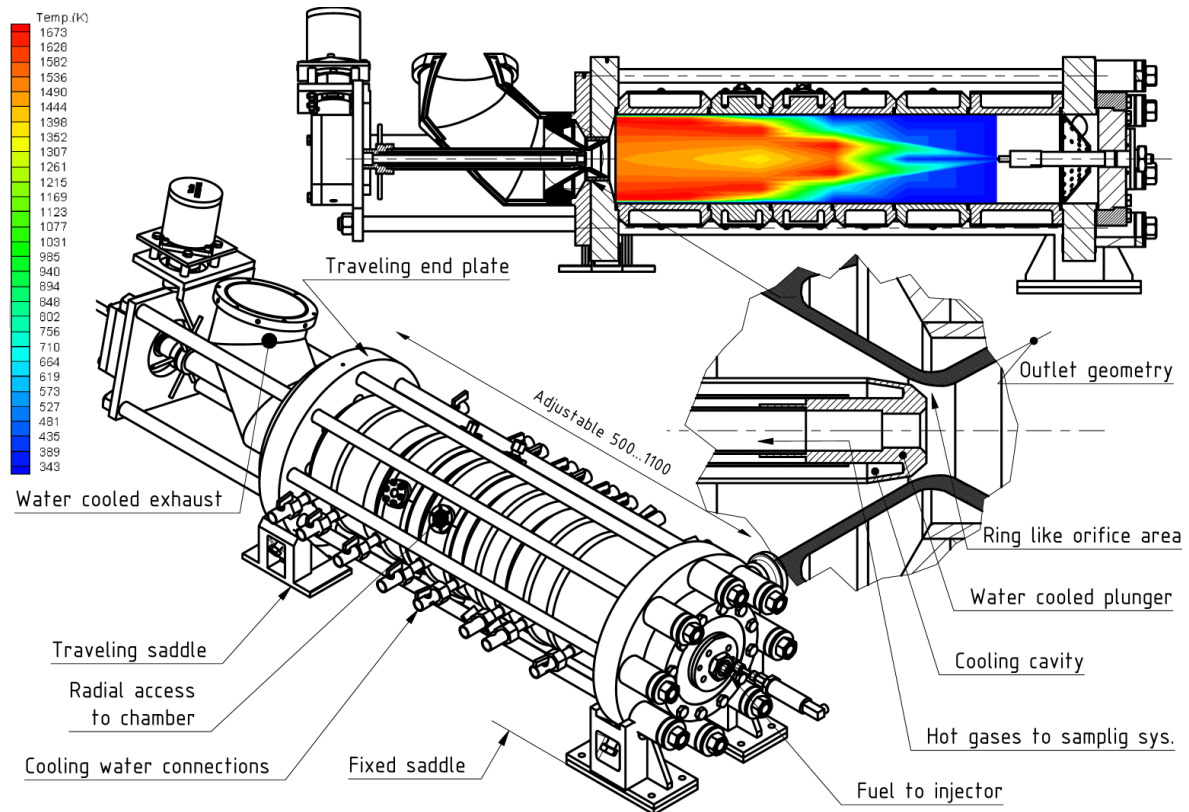
281 **3.6 Other design considerations**

282 In this section further design considerations are listed and referenced. They are not of interest to this study, but they can
283 be potentially important for other designs, for instance, in applications where the pressurised chamber is used with high
284 accuracy polarisation or interference instruments or deep space applications.

- 285 • Compressive stress caused by sharp edges on the surface of an optical element [78, 79]
- 286 • Effect of a temperature gradient on adhesive bonds [52, 80]
- 287 • The tensile stress in a brittle material due to a compressive load on its surface [52, 63, 81]
- 288 • Focus shift in thick parallel plane optical elements [82]
- 289 • Distortion caused by a temperature gradient [71, 83-85]

290 **4 Design of an optically accessible pressure chamber**

291 The detailed geometry of the existing combustor is shown in Figure 9. An important feature of it is that the working
292 chamber was constructed from sections. Utilising this property, the reactor's length could be varied to adjust the
293 residence time of the reactants. Different length sections were available to build the reactor. When the reactor was
294 assembled, the sections were sealed by polymer O-rings. As a result of the limited temperature resistance of the stainless
295 steel structure and the high thermal load, the sections had to be individually water-cooled, as in Figure 9.



296

297

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Figure 9. An isometric and a section view of existing chamber, computational result of in-cylinder temperature distribution shown in the section view. Conditions: diesel fuel; stoichiometric ratio; 6 bar in-reactor pressure.

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4.1 Section

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4.1.1 Design requirements

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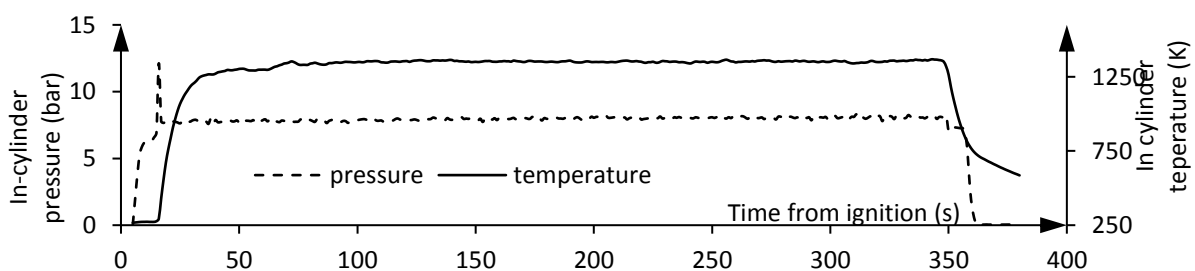
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It was required that the optical section withstand the maximum of 20 bar working pressure at the maximum possible operating temperature and that its geometry would allow it to be connected it to the existing rig. In order to seal the reactor, it was essential to maintain the temperature at an acceptable level in the O-ring grooves. The maximum continuous operating temperature of the Viton O-rings (200 °C) was chosen as a limit on the surfaces that were in contact with the rings. The system could provide a maximum cooling flow rate of 10 litres per minute for the new optical section. The highest allowed inflow cooling temperature was 70 °C. The estimation of the heat flux coming from the combustion to the section was based on a number of test results where an in-chamber, single-point gas temperature measurement was taken. An example of these results is shown in Figure 10. The results of computational work on combustion and in-chamber conditions by Demosthenous and Crookes [10] were used as input boundary conditions for the analysis, Figure 9. It was also a requirement that the windows could be easily changed to metal blanks for heating up or non-optical tests.



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Figure 10. Typical test results of the existing combustion chamber; the time of ignition is indicated by the rapid increase in temperature and a pressure peak at around 10s after ignition

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It was an underlying requirement that the new apparatus would allow investigation with a Phantom 4.3 high speed camera, a TSI Particle image velocimetry system (laser: dual 50 mJ/pulse, NewWave Gemini Nd:YAG; sensor: PowerView 4MP) and a FLIR Titanium 560M infrared (IR) camera.

317

4.1.2 Material choice

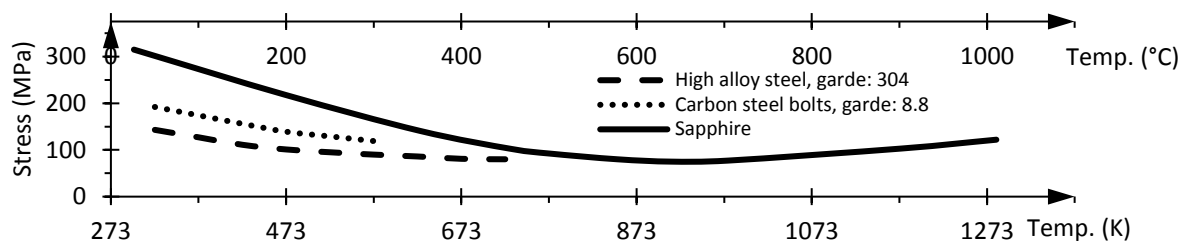
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As shown in Figure 12, the resultant optical section is a complex shape, featuring fine finished surfaces for sealing purposes. It is thus that the material of the section body needed to have adequate strength to withstand the pressure load at high temperatures. It also needed to be suitable for precision subtractive and additive manufacturing. Moreover, due to

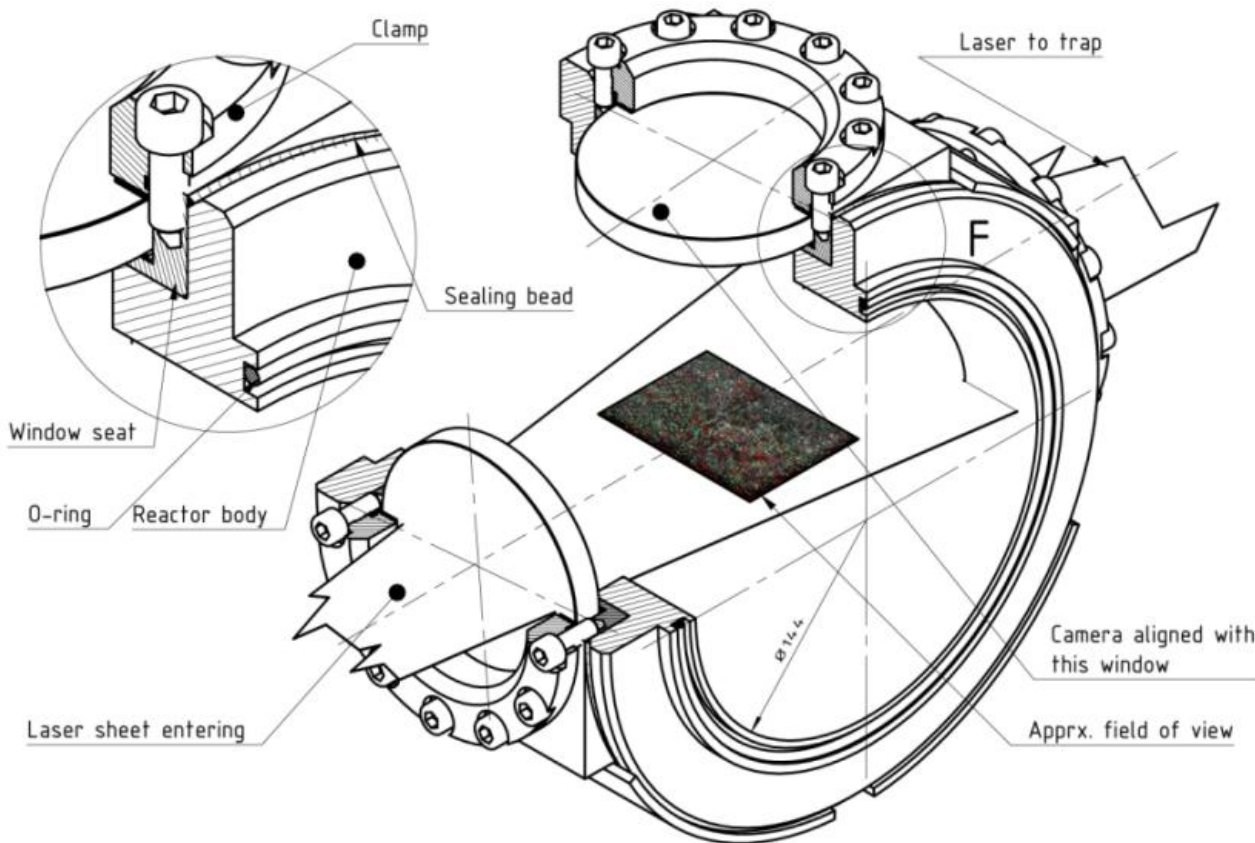
321 the corrosive products forming inside the chamber and the constant presence of cooling water, the material was required
 322 to have some corrosion-resistant properties. A detailed list of possible materials can be found in EN134453 [86]. The
 323 aforementioned requirements suggested using an austenitic stainless steel grade. After considering the cost, the corrosion
 324 resistance and manufacturability grade 304 (1.403) was chosen. It is easily available, with well-documented data on its
 325 mechanical properties at elevated temperatures showing the tensile stress for the materials that were used to construct
 326 the optical combustor, Figure 11.



327
 328 **Figure 11. Permissible design stresses of some selected materials as function of temperature [37, 56].**

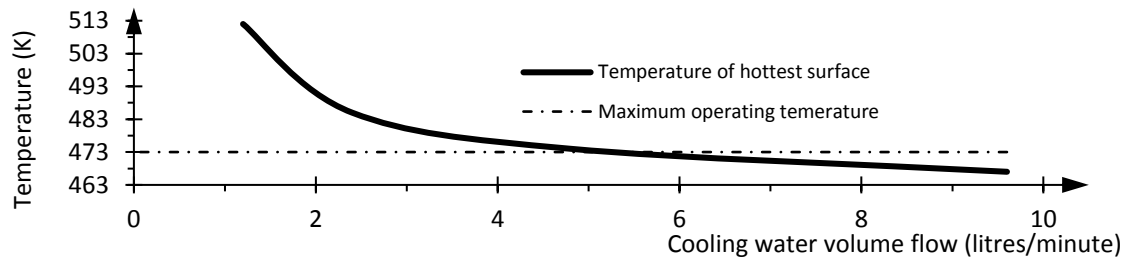
329 **4.1.3 Design of required geometry and validation**

330 This structure of the overall design defined some underlying properties of the optical access. It was the obvious choice
 331 that the optical access should be constructed on a section which has similar dimensions and had the same sealing method.
 332 Given the nature of the laser-radiation-based measurement method, a three-access point – in a T-like configuration – was
 333 required. A detailed review of optical measurements can be found in the books by Zhao [12, 13]; see details in Figure 12.
 334 This three-access point design was satisfactory for the high speed and IR camera setup.



335
 336 **Figure 12. The final design of the optical section. The laser sheet entering and leaving the chamber is also indicated.**

337 It can be seen that cooling was essential for the operation of this reactor. Insufficient cooling would result in a rapid
 338 increase of temperature in the body and in the window (or blank). With the increased temperature the polymer seals
 339 would quickly fail. Therefore, the geometry of the body with window seats had to provide enough surface area for the
 340 coolant and allow sufficient volume flow. Assuming the largest heat flux and inflow cooling temperature, a number of
 341 simulations were carried out to plot the O-ring groove temperature curve against the coolant flow rate. It was found that
 342 having at least 5.5 litres per minute coolant flow rate on the designed geometry could keep the O-ring groove
 343 temperature at an acceptable level. The 5.5 litres per minute minimum cooling flow requirement is below the 10 litres per
 344 minute maximum performance, therefore the given cooling system was found to be sufficient.



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Figure 13. The O-ring groove temperature as the function of the coolant volume flow (value was taken from the corner point of the hottest possible cross section, as indicated in Figure 15b).

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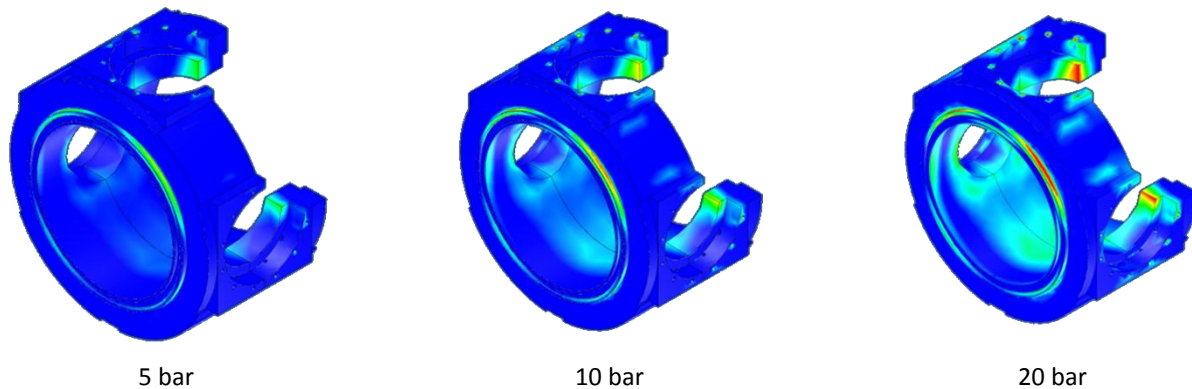
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The length of the section was chosen to be the same as that of the longest existing section. The thickness of the optical section was based on the existing design. Polymer O-rings were used to seal between the sections, the design and manufacturing were according to BS ISO 3601 [87]. With the given length, a circular window type was selected for easier machining and the geometrically maximum possible diameter, 82 mm, was chosen to be evaluated. FEA was carried out to determine the stress arising from thermal loads and applied pressure. As is indicated in Figure 14, the highest stresses are in the O-ring grooves and in the openings. The grooves had high stresses on their contact surfaces because of the large axial force pressing the sections together against the pressure. In the case of the openings the high stress rate can be explained by the reduced material volume, i.e. reduced inertia [88].



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Figure 14. Stress distribution as a function of chamber pressure

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The inner diameter of the optical section is 144 mm; the outer diameter is constrained by the support rods, as seen in Figure 9. As the radial space was limited, the best design solution would have been to place the optical element inside the shell – but with the given casted base, this was not achievable. Therefore, the second best option – with an eye to optimising storage space – was to have a free sitting clamp; this setup is shown in Figure 12. The detailed section-view shows the window kept in place by the circular clamp. The clamps were fixed and positioned by 12 M6 socket-head bolts to the window seats. The window seats were welded all around to the base chamber or reactor body; the beads sealed the seats and kept them in place. The window to seat-sealing surfaces were precision-manufactured, as suggested in PD5550:2009 Table 3.8-3 [89]. With the aid of the measurements and computational results, further analysis was carried out to estimate the working temperature of the section and optical elements. The temperature distribution of the cross-sections is shown in Figure 15.

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4.2 Window

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4.2.1 Design requirements

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The main purpose of the window is to provide a transparent barrier between the combustion and instrumentation. An underlying requirement was to allow use of three different sensors (cameras) with differing spectral responses. Wavelengths of interest are particulate imaging velocimetry (around 532 nm), high speed camera (visible spectra) and IR camera (3 to 5 micron); the responsivity curves are shown in Figure 2. As the wavelength of the laser is in the visible range, there were two bands of electromagnetic radiation that needed to be considered – 380-985 and 2800-5200 nm. In these regions, the minimum of 80% transmittance was required. The tolerance on the OPD had to be kept in the photographic range: 10 nm/cm. As the three sensors were robust and the measurements by them were not overly sensitive, the maximum allowed lens power of the distorted window was 10^{-6} dioptre. The maximum expected pressure difference on the optical element was 20 bar. The required safety factor was required to be four for stresses arising from mechanical and thermal loads, with a maximum probability failure of 10^{-4} . The technological considerations and tolerances were

379 chosen to fit laser and the precision measurements requirements. The level of precision was selected to be good
 380 according to Table 3. Finally, the operating temperature of the window had to stay under the maximum permitted level.

381 4.2.2 Material choice for optical element

382 The spectral requirement is shown in Figure 2 along with the transmittance curves. The ideal design solution was to select
 383 only one material type to cover the required wavelength ranges. It can be seen that the two possible material types that
 384 cover the needed large range of EM wavelengths are sapphire and magnesium fluoride. The thermal analysis of the optical
 385 section indicated that the steel blanks and windows would need cooling to survive. The calculations and simulations were
 386 carried out for both materials. It was found that MgF₂ can be a valid option for low-temperature and low-humidity
 387 environments. Extra caution is required when a temperature gradient is applied on the MgF₂ material, as its high
 388 expansion coefficient and middle-range conductivity combined with low strength makes it sensitive to thermal shock. It is
 389 also suggested by manufacturers that MgF₂ can react with high temperature steam similar to the one that can be found in
 390 the HPC as a combustion product. Therefore, sapphire was, instead, chosen as material for the windows as it combines
 391 good transmittance in all the required wavelength bands as well as having good thermo-mechanical strength.

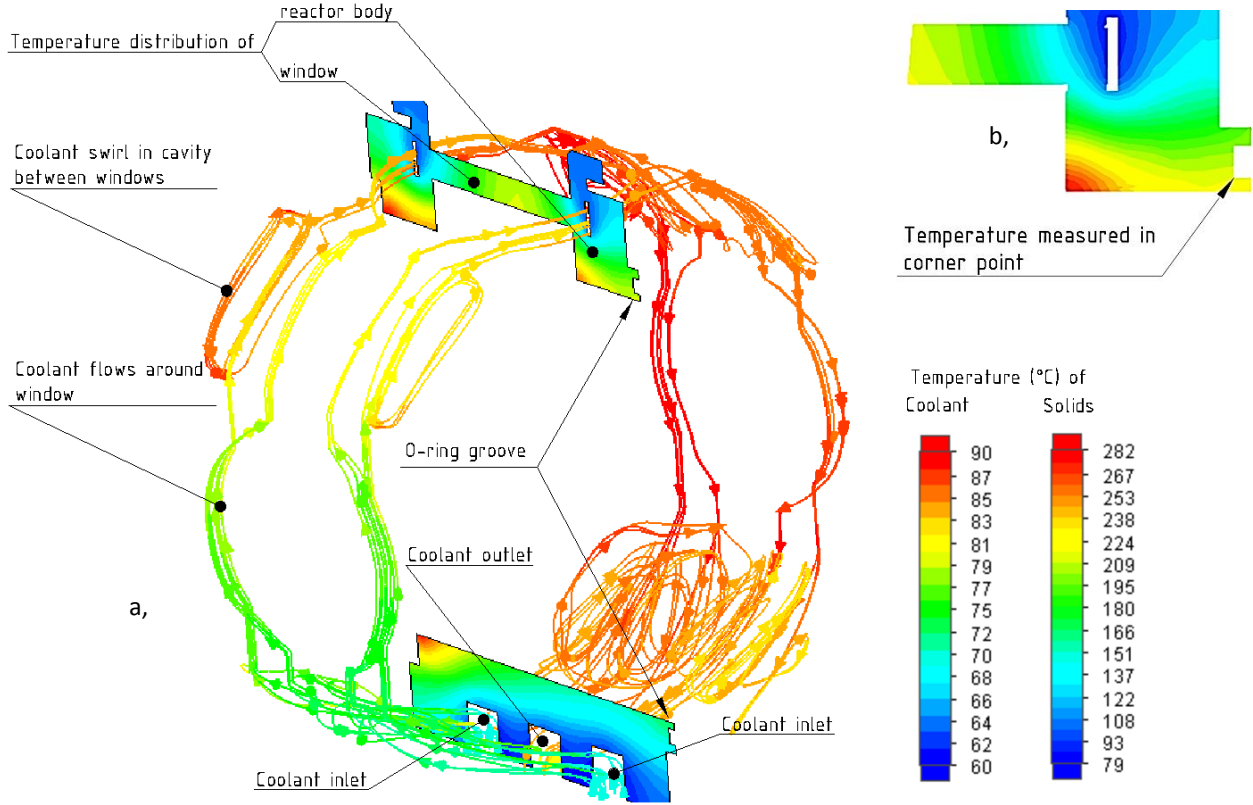


Figure 15. a) An isometric view of the flow of cooling water, its temperature change and the temperature distribution of a cross section of the reactor. b) Temperature distribution around the O-ring groove.

392 4.2.3 Design of required geometry and validation

393 The diameter of the window was determined by the maximum available space in the optical section. The maximum
 394 possible diameter was found to be 82 mm. Previous experience showed that a width of 9 mm minimum contact surface is
 395 required to provide an adequate sealing performance. The 9 mm wide contact ring also ensured an acceptable level of
 396 compressive stress in the window and provided large enough heat transfer surface for the metal blanks. This geometric
 397 design resulted in a 64 mm aperture. The thickness of the window was estimated by using the equations and relations that
 398 were explained earlier in this paper, and then validated by FEA. The minimum thickness was calculated for two
 399 requirements: maximum allowable stress and OPD. Substituting values to Equations (11) and (17) the thickness value
 400 results were 5.88 and 0.07 mm respectively.

$$t_{min|\delta} = \left(\frac{1}{2}D_o\right) \left[\frac{K_w S F_{\delta} \Delta p}{\delta_{des}}\right]^{1/2} = \left(\frac{1}{2}64\text{mm}\right) \left[\frac{0.75 \cdot 4 \cdot 2\text{MPa}}{178\text{MPa}}\right]^{1/2} = 5.88\text{mm} \quad (18)$$

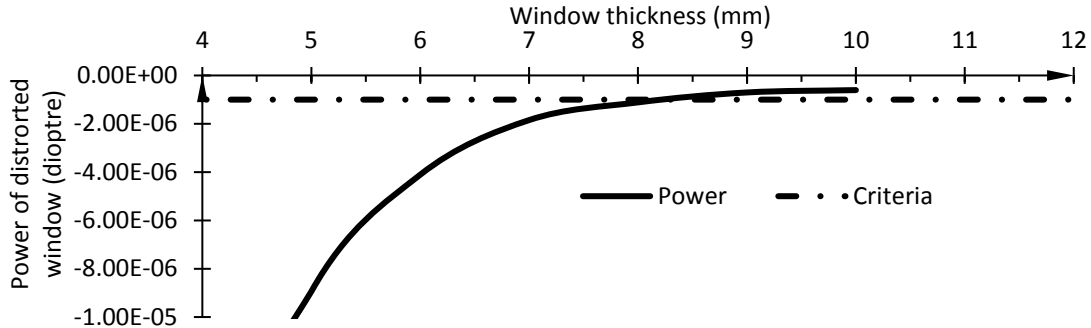
$$t_{min|ODP\Delta p} = \sqrt[5]{8.89 \cdot 10^{-3}(n-1) \frac{\Delta p^2 D^6}{OPD \cdot E^2}} = \sqrt[5]{8.89 \cdot 10^{-3}(1.76-1) \frac{(2\text{MPa})^2 \cdot (64\text{mm})^6}{10 \frac{\text{mm}}{\text{cm}} (345\text{GPa})^2}} = 0.07\text{mm} \quad (19)$$

401 The results of the estimation indicated that, with the given loads and geometry, the required optical performance was
 402 easily achievable. Then, the window deflection was calculated using FEA for the highest thermal and mechanical loads. The

403 stress and deflection results were substituted in Equations (13), (14) and (15) in order to check the design for failure
 404 probability and image distortion. Equations (13) and (14) combined together gives the power of a distorted window as
 405 function of refractive index, aperture and deflection.

$$P_{lens} = \frac{64(n-1)tx_{(t)}^2}{(D_0^2 + x_{(t)}^2)(D_0^2 + x_{(t)}(x_{(t)} - 8t))} \quad (20)$$

406 The power as function of window thickness for the given geometry is shown in Figure 16.



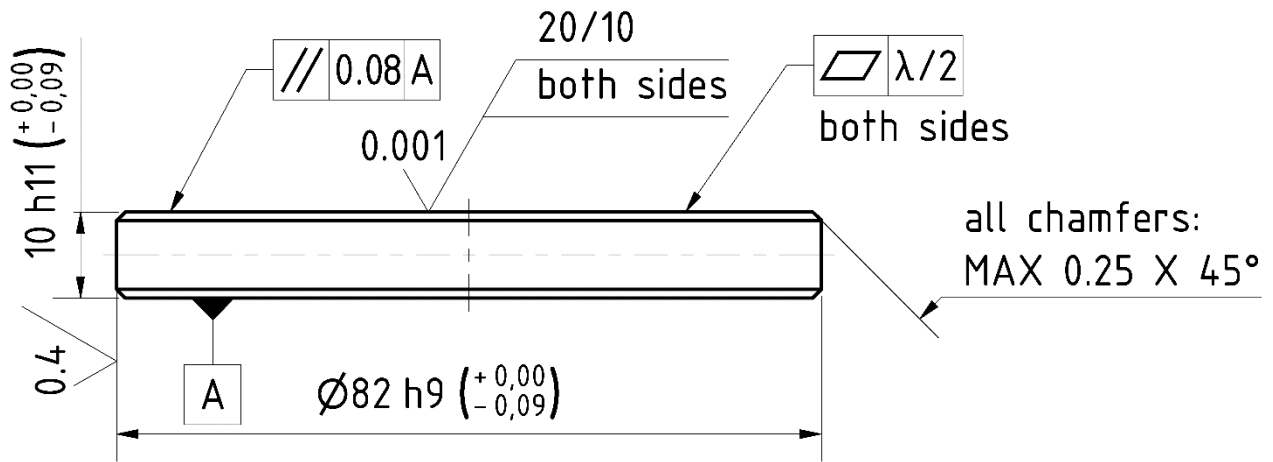
407
408 **Figure 16. Power of distortion vs window thickness**

409 It was found that the limiting factors were the probability failure and deflection. Based, on the curve above and financial
 410 consideration the thickness was chosen to be 10mm. Using FEA, the maximum stress was found to be 42.8MPa, with this
 411 level of stress:

$$P_f = 1 - \exp\left[-\left(\frac{\sigma}{\sigma_0}\right)^m\right] = 1 - \exp\left[-\left(\frac{42.8\text{MPa}}{485\text{MPa}}\right)^5\right] = 5.35 \cdot 10^{-6} \quad (21)$$

412 However, there is a high inaccuracy in the result of probability failure and therefore the result is only a guideline. The
 413 value of σ_0 is function of a number of variables, one of the most important ones is temperature. There is a no data
 414 available for σ_0 in the literature for elevated temperatures.

415 **5 Results and Conclusion**



416
417 **Figure 17. Side view: final sapphire sight window design, where $\lambda = 532\text{nm}$**

418 This work has investigated the design of pressure vessels equipped with optical access. A significant number of data
 419 sources were surveyed to produce a comprehensive review of the most related optical, thermal and mechanical
 420 properties for some optical materials. The database was presented in a way that allows for a convenient and direct
 421 comparison. The little design data that was available on the topic in the literature was presented, while also considering
 422 practical, mechanical and optical design considerations. The aspects of the design for optical performance were described
 423 in-depth, with additions to the already published equations and relations. The utilisation of this database allowed for the
 424 design process of a pressure chamber, with optical element under high mechanical and thermal load, to be demonstrated,
 425 where, the required experimental rig needed to support research activity for a range of optical instrumentation. Figure 18
 426 indicates the results gained from the working optical section.

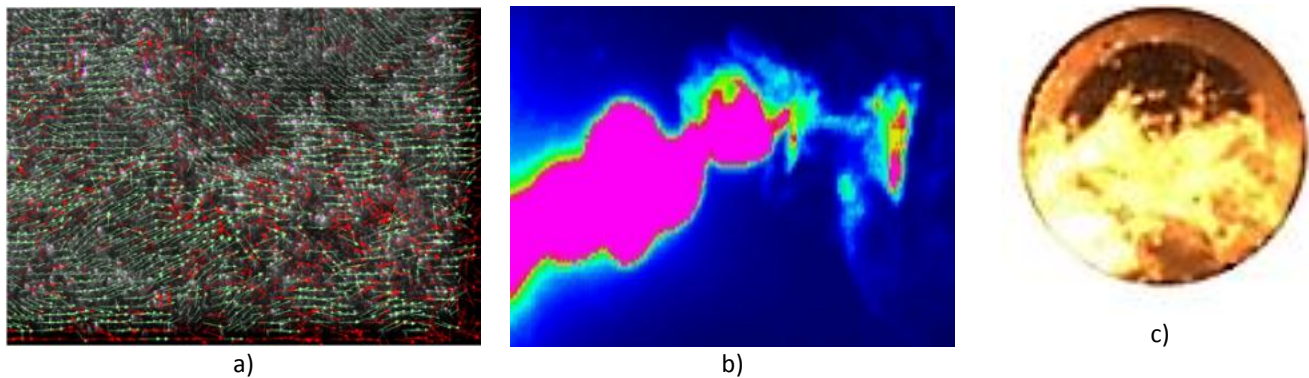


Figure 18. Results from the optical HPC: a) PIV vector field of the flowing air fuel mixture. b) IR radiation image recorded from a hydrocarbon flame. c) A still image from high speed video.

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